

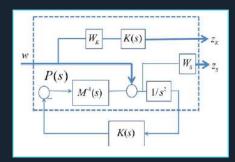


RESEARCH ACTIVITIES – <u>AGILE AOCS</u>

- □ AOCS Attitude control for agile and flexible systems with specific focus on large flexible antennas
 - Remote sensing satellites require agility to improve revisit performances theater imaging.
 - However radar satellites are equipped with large antennas requiring special AOCS design (for example H_{∞} Design with Acceleration Sensitivity).



- AOCS for flexible structures -



- Mixed acceleration sensitivity set up -(image taken from "Flexible Spacecraft Dynamics, Control and Guidance" by L.Mazzini)

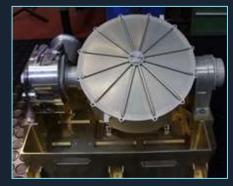






RESEARCH ACTIVITIES - AGILE AOCS

- □ Agility requires the development of new control actuators and specific AOCS structures
 - □ Control Moment Gyro (CMG), an actuator able to strongly enhance satellites attitude agility.
 - **Miniaturized Control Moment Gyro** (MCMG) where electronics and mechanics have been optimized for the use on microsatellites.
 - Gyro-CMG internal loop control Large bandwidth Gyro-CMG loop to realize high fidelity target acquisition profiles with minimal effect from flexible modes



- Control Moment Gyro -



- Mini Control Moment Gyro -



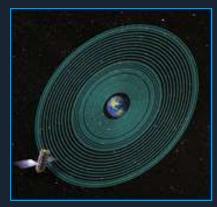




RESEARCH ACTIVITIES - ORBITAL TRANSFERS OPTIMIZATION

□ Orbital transfers optimization

- The use of *plasmic propulsion* allows telecom satellite transfer from **LEO to GEO** orbit, we are developing an electrical platform (*ItalGovSatCom* with ASI) compatible with **VEGA** injection in LEO.
- The Optimal Steering of the thrust direction requires the use Optimal
 Control Theory applied to very long low thrust orbital transfers
 - ✓ Hamilton-Jacobi-Bellman equation with averaging
 - ✓ Sufficient and necessary conditions for non-regular Hamiltonian
 - ✓ Systems of autonomous navigation based on characteristics method



- Very long orbital transfer -



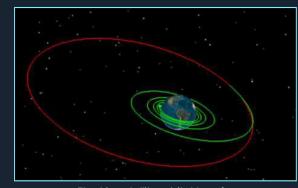
- Autonomous station keeping for IOS -



RESEARCH ACTIVITIES – <u>ORBITAL TRANSFERS OPTIMIZATION</u>

Hamiltonian technics for Orbital Transfers Optimization of electrical satellites are used to:

- Calculate on board the LEOP transfer steering laws,
- Calculate on board orbit keeping maneuvers using <u>GPS receivers in GEO</u> which allow also precise satellite collocation.
 - Electric propulsion system implies very <u>low thrust</u> thus the orbital transfers may take <u>many</u> <u>months</u> and hundreds of orbits requiring specific averaged optimization.
 - ☐ To reduce the operational cost of the LEOP **AUTONAV** function allows to compute on board the optimal trajectory in real time, avoiding several ground segment contacts.

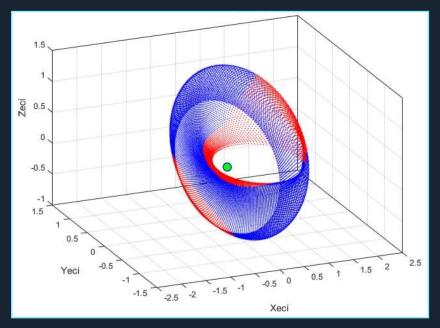


- Electric satellite orbital transfer -



RESEARCH ACTIVITIES - ORBITAL TRANSFERS OPTIMIZATION

Low thrust orbital transfer minimizing the mass propellant consumption



- 3D trajectory for a GTO to inclined MEO (56°) in Vmin mission analysis where the thrusters are switched off (blue) and switched on (red)-





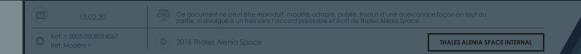


PLATFORM DEVELOPMENT - <u>MICROSAT PLATINO</u>

- □ **PLATINO** is a Micro multi-application platform capable to respond to the international and Institutional market opportunities (P/L mass up to 80 kg) it is a project in collaboration with Sitael
- The platform electronics is based on an highly integrated design and the use of **COTS** in order to get low masses and costs
- Only three electronics
 - ☐ The **IPAC** for all Avionic and Power functions
 - ☐ The ICU for the transmission to ground
 - ☐ The **mCMG** for the actuation control



- PLATINO -

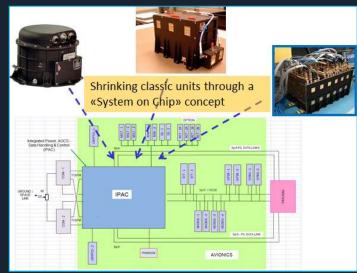




PLATFORM DEVELOPMENT - <u>PLATINO KEY TECHNOLOGIES</u>

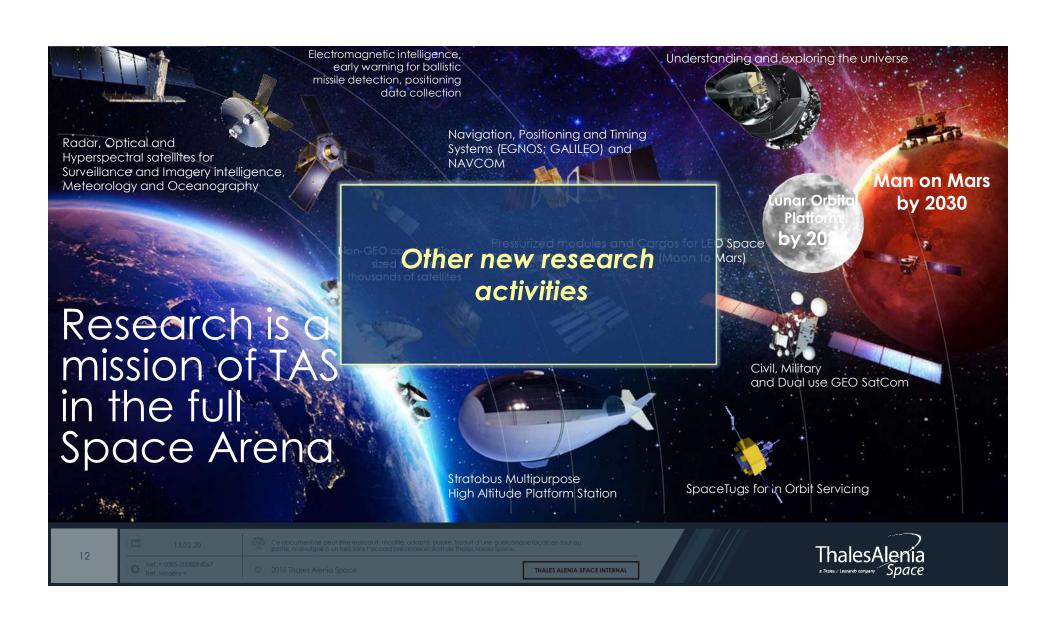
Integrated Avionics with **IPAC** (Integrated Power, Attitude Control And Data Handling Control Unit):

- ☐ The IPAC contains the following functional and fully redundant modules:
 - Multi Core Processor Module (MCPM)
 including Star Trackers and GPS processing
 - Telemetry, Telecommand, Mass Memory
 - Reconfiguration Module (T2MR)
 - Input & Output Module (IOM)
 - Power Module (PWM)









OTHER NEW RESEARCH ACTIVITIES

- Large Deployable Reflector (LDR) for Radar & NAVCOM & TLC solutions Improving P/L size with mass optimization
- Insert new digital technologies in space-borne radars and Communication evolving space-borne radar capabilities thanks to Digital Beam Forming and introducing Post 28nm Deep Sub Micron ASIC Technologies in Radar and TLC
- **Quantum Communications** Architecture & Critical elements (QKD) defining the architecture of satellite Quantum Communication system
- ☐ Innovative solution for Exploration & Space Commercialization, habitative pressurized modules for human operations in space (SpaceHOME), platform for In-Orbit Services

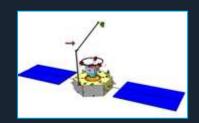
(SpaceSTART), landers and re-entry vehicles, robotics, rovers for planet exploration, etc.)



- Large Deployable Reflector -



- Opticla feederlinks for VHTS -



- In Orbit Services -





- SpaceHOME -



OTHER NEW RESEARCH ACTIVITIES

- Evolution of Digital technologies and porting to Space Missions
 - Big Data, AI and Internet of Things
 - ✓ Al algorithms for Image Processing and Image Analytics
 - ✓ Al on Exploration Rovers autonomous navigation
 - Cybersecurity in four domains
 - ✓ Processes of production,
 - ✓ On Board S/W
 - ✓ Ground Segment
 - ✓ Satellite Ground segment link







