A Constellation of CubeSats for Amazon Rainforest Deforestation Monitoring

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Mission Goals

Requirements

Orbit Char acteristics

Payload

Satellite Bus

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Amazon Rainforest

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- The Amazon represents over half of the planet's remaining rainforests.
- The largest and most biodiversity tract of tropical rainforest in the world.
- Actually is monitored by brazilian government using a free database.



Figure: Limits of Amazon rainforest.





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Figure: A comparison between the years 2002 and 2017.





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- Develop a new system to identify new methods of deforestation in Brazilian Amazon Rainforest.
- Advance the capability of University of Brasília to design, develop and operate small satellites.
- Provide educational opportunities related to aerospace missions, satellites design and project management.







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- A functional spacecraft shall be designed.
- Minimum spatial resolution less then 16 meters.
- RGB and NIR range.
- Maximum revisit time of 16 days.
- Images taken from the same place shall have similar illumination conditions.
- Low cost mission





Orbit Characteristics

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• The resultant orbit is a sun-synchronous circular orbit.

• Will be necessary 10 satellites in the constellation, that provides 13 days of revisit time.

Table: Orbital characteristics.

Altitude	500 <i>km</i>	
Inclination	97.40 <i>degrees</i>	
Orbital Period	95.55 <i>min</i>	
Swath	7.7 <i>km</i>	
Spatial Resolution	15 <i>m</i>	
Orbital Velocity	7.53 <i>km/s</i>	





Payload

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 The selected payload is a Multispectral Camera -Red/Green/Blue/NIR - USB3, from Spectral Devices, that attends all requirements.

Table: Specifications of the payload

Interface	USB3	
Maximum Bit Depth	12 bits	
Number of Channels	4 bands	
Pixels Per Channel	512 <i>×</i> 512	
Dimensions	52 <i>x</i> 46 <i>x</i> 56 <i>mm</i> ³	







Payload

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Table: Results in the calculation of data generated

Data Generation Characteristics				
V_{g}	6.98 <i>km/s</i>			
D_g	7.65 <i>Gb/orbit</i>			
Ň	15.07 <i>orbits/day</i>			
D_T	115.29 <i>Gb/day</i>			

as V_g = relative ground velocity D_g = generated data per orbit N = number of orbits per day D_T = generated data per day





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Figure: CubeSat 3U





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- The reaction wheel selected, based on the sum of disturbance torques from table, was CubeWheel Small from CubeSpace.
- A magnetorquer will be used to de saturate the reaction wheels and de tumbling. We selected a magnetorquer board from ISIS.

Table: Disturbance Torques acting on the CubeSat.

Disturbance Torque	Magnitude (Nm)	
Gravity Gradient	$2.1356 \cdot 10^{-8}$	
Solar Radiation	$2.1872 \cdot 10^{-10}$	
Aerodynamic	$1.7386 \cdot 10^{-7}$	
Magnetic Field	$9.7856 \cdot 10^{-6}$	





Actuation System

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• This system will be composed by solar panels as sources, rechargeable batteries, a battery charger regulator and a system distribution.

Table: Power Budget

	Component	Cycle %	Peak (mW)	Average (mW)
	Radio Receiver	100	5500	5500
us	Radio Transmitter	5	11000	550
	Radio Beacon	100	300	300
	OBC	100	660	660
	Payload	5	4500	225
s	Reaction Wheels	30	600	180
ı s	Sun Sensor	100	13.4	13.4
	Magnetorquer	10	1200	120
	Thruster	10	40000	4000
	Total		= 63773.4	= 11548.4



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- More detailed description of subsystems.
- Launch and Ground Station selection.
- Complete documentation of phase B.
- Funding to initiate phase C.





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- Nowadays monitoring satellites can not identify the new trend of deforesting small areas.
- 10 CubeSats in a constellation will be enough to achieve requirements.
- Revisit time of 13 days and spatial resolution of 15m.
- The project presented here was just pre phase A, phase A and beginning of phase B.





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Thank You!



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